

1. Start XFOIL (double-click "xfoilP4.exe").
The prompt should be "XFOIL c>"
2. Load the file "clarkyl4sf.dat"
3. Type "PPAR" to view airfoil contour/nodes.
Hit "Enter" in command window to return.
4. Enter .OPER menu by typing "oper"
The prompt should be ".OPER: c>"
5. Default is inviscid mode. Change to viscous:
Enter "v," then type the value for Re.
Enter "m," then type the Mach no.
The prompt should be ".OPERv c>"
6. Set iteration limit : Enter "iter" then type "100"
7. Adjust BL calculations to account for a
slightly "dirty" tunnel, like the ENT:
Enter "Vpar," then type "n," then type "7."
Hit "Enter" twice.
8. To run an ACA value, type "alfa 0" or "alfa 5", etc.
9. Save the plot hardcopy: type "hard"
10. Save the Cp data by typing "cpwr"
Enter a filename like "AOA00.dat" or "AOA05.dat"
11. When done, hit "Enter" to return to the
main menu ("XFOIL"). Then type "quit."
12. Turn the plot file "plot.ps" into a PDF by
using Acrobat Distiller (just double-click
the file).

This will
append all
plots into
the same
file.